

DIRECT STRENGTH PREDICTION OF LIPPED CHANNEL COLUMNS EXPERIENCING LOCAL-PLATE/DISTORTIONAL INTERACTION

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ABSTRACT: This paper reports an ongoing investigation intended to assess the performance of the Direct Strength Method (DSM) to estimate the ultimate strength of lipped channel columns affected by local-plate/distortional mode interaction. First, the DSM approaches to safety check columns against local-plate and distortional failures are briefly reviewed, with special attention devoted to a recently proposed extension that takes into account the above buckling mode interaction. Next, one presents and discusses the results of a parametric study, carried out by means of the code ABAQUS and involving the evaluation of the “exact” ultimate loads of 63 lipped channel columns with various geometries, all exhibiting local-plate/distortional interaction. Then, these ultimate strength data are compared with the estimates provided by the available DSM formulae and, on the basis of this comparison, one identifies several features that a DSM approach successfully accounting for local-plate/distortional interaction must incorporate.

Keywords: Direct Strength Method (DSM), cold-formed steel, lipped channel columns, strength, local-plate buckling, distortional buckling, local-plate/distortional interaction

1. INTRODUCTION

The Direct Strength Method (DSM) was originally proposed by Schafer and Peköz [1], about nine years ago, and has been continuously improved since, mainly due to the research activity carried out by Schafer [2, 3]. Moreover, one should mention that the inclusion of the DSM in the AS/NZS and NAS specifications for cold-formed steel design has been very recently approved – they already appear in the current (new) versions of these codes [4, 5]. The method has been shown to provide an efficient general approach to estimate the ultimate strength of cold-formed steel columns and beams (i) exhibiting global (flexural, torsional or flexural-torsional), distortional or local-plate failure modes or (ii) failing in mechanisms that involve interaction between global and local-plate buckling modes. Indeed, the most recent DSM version stipulates the need to perform two independent safety checks, regardless of the member critical buckling mode nature: (i) one against a distortional failure and (ii) the other against a local-plate or a combined local-plate/global collapse. In the latter case, the DSM provides an efficient alternative to the more traditional and conservative “effective width method”. However, as pointed out by Schafer [2, 3, 6, 7], further research is needed before the DSM approach can be successfully applied to members (i) under compression and bending [8, 9] or (ii) influenced by interaction phenomena involving distortional buckling modes [10-13]. Since it has been recently shown that the post-buckling and ultimate strength behaviours of various commonly used lipped channel cross-section shapes can be strongly affected by coupling between local-plate and distortional buckling modes [14-16], it would be obviously very convenient to have this mode interaction phenomenon also covered by the DSM.

The objective of this work is to contribute towards the extension of the domain of application of the available DSM, by making it able to estimate the ultimate strength of lipped channel columns affected by interaction involving local-plate and distortional buckling modes. In order to achieve this goal, one begins by presenting the main results of a shell finite element investigation (performed with the code ABAQUS [17]) concerning the elastic-plastic post-buckling behaviour (up to collapse) of lipped channel columns experiencing local-plate/distortional buckling mode interaction. Then, one reports an extensive parametric study involving the determination of the elastic-plastic failure loads of lipped channel

columns with distinct cross-section dimensions, lengths and yield stresses, and containing critical-mode (distortional) small-amplitude initial geometrical imperfections – the columns were carefully selected, in order to exhibit strong local-plate/distortional interaction effects. All second-order elastic-plastic analyses were performed in the finite element code ABAQUS and discretising the columns into 4-node shell elements. These ultimate strength values then provide a “data bank” that makes it possible to propose and validate preliminary recommendations concerning the use of a DSM approach to estimate collapse loads of columns affected by local-plate/distortional mode interaction – as already mentioned, the results reported here deal with an ongoing investigation aimed at proposing (slight) modifications to the existing DSM equations, thus making them applicable to the design of lipped channel columns against this mode interaction phenomenon.

2. LOCAL-PLATE/DISTORTIONAL BUCKLING MODE INTERACTION

One reports here the main results of a recent investigation on the elastic-plastic post-buckling behaviour and ultimate strength of simply supported (plain) lipped-channel columns that are strongly affected by local-plate/distortional buckling mode interaction – the columns analysed (i) have the cross-section dimensions, length and elastic constants indicated in Figure. 1(a) and (ii) buckle elastically for $\sigma_{cr}=100.5$ MPa in arbitrary combinations of (ii₁) a 3 half-wave local-plate mode and (ii₂) a single half-wave distortional mode, as illustrated in Figure. 1(b) [16]. The ultimate strengths were obtained through *finite element* analyses (FEA) carried out in the code ABAQUS [17] and adopting *shell* elements to discretise the columns. As far as the performance of these FEA is concerned, the following aspects deserve to be mentioned here [14, 18]:

- (i) Discretisation. The column mid-surfaces were discretised into S4 finite elements (ABAQUS nomenclature: isoparametric 4-node shell elements with the shear stiffness yielded by a *full* integration rule), which were found to be the most adequate to carry out this task. One considered 20-30 elements along the cross-section mid-line (width of about 10 mm) and previous convergence/accuracy studies showed that the finite element length-to-width ratio should be comprised between 1 and 2.
- (ii) Support Conditions. All columns have end sections locally/globally pinned and free to warp. Concerning the first aspect, these support conditions were modelled by imposing null transverse membrane and flexural displacements at all end section nodes – in order to preclude a spurious longitudinal rigid-body motion, the axial displacement was prevented at one mid-span cross-section node.
- (iii) Loading. Compressive forces, statically equivalent to a uniform normal stress distribution, are applied at the nodes of the column end-sections. Since the reference value of the *load parameter* p is t N/mm (t is the wall thickness), which corresponds to a 1 MPa uniform stress distribution, the value of p yielded by ABAQUS is numerically equal to the *average stress* acting on the column (expressed in MPa).
- (iv) Material Modelling. The column (carbon steel) material behaviour, deemed isotropic and homogeneous, was modelled through (iv₁) linear elastic (bifurcation analysis) and (ii) elastic/perfectly-plastic (post-buckling analysis) stress-strain laws. In the latter case, the well-known Prandtl-Reuss model (J₂-flow theory), which combines Von Mises’s yield criterion with the associated flow rule, was adopted. These stress-strain laws are readily available in the ABAQUS material behaviour library and one just needs to provide the values of E , ν and f_y – one considered $E=210$ GPa (Young’s modulus), $\nu=0.3$ (Poisson’s ratio) and five different yield stresses, which correspond to yield-to-critical stress ratios equal to $f_y / \sigma_{cr} \approx 1.2, 2, 3.5, 5.5$ (in elastic columns, included here for the sake of completeness, one obviously considered $f_y / \sigma_{cr} = \infty$).

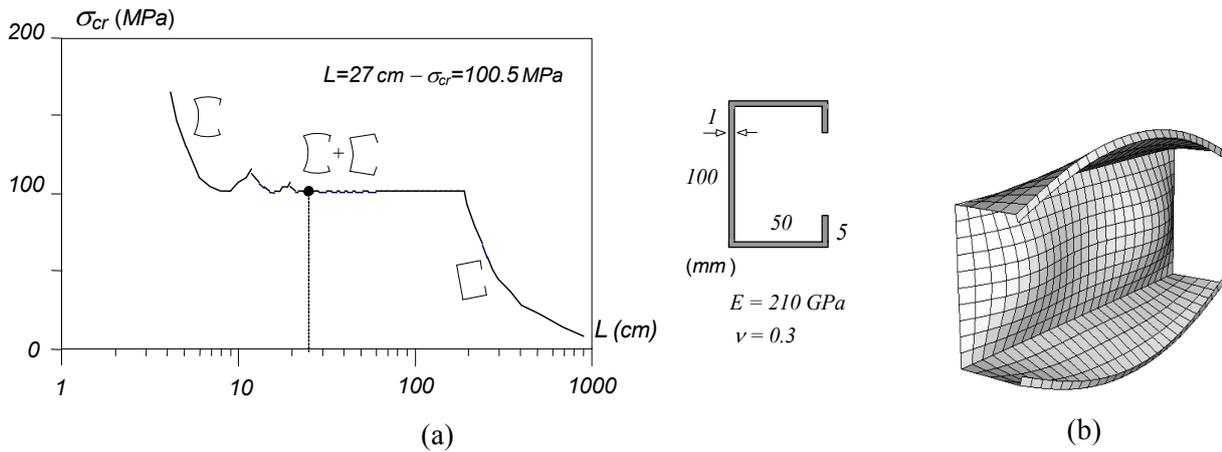


Figure 1. Buckling Results: (a) σ_{cr} vs. L Curves (b) Critical “Combined” LP/D Mode Shape

2.1 Initial Geometrical Imperfections

The shape of the initial geometrical imperfections plays a crucial role in mode interaction investigations, since its choice may alter considerably the post-buckling behaviour and ultimate strength of the structural system under consideration. Indeed, the usual approach of including critical-mode imperfections ceases to be well defined, due to the presence of two “competing” buckling modes that may be combined arbitrarily – in this case, a three half-wave local-plate and a single half-wave distortional buckling modes. Thus, in order to obtain column equilibrium paths that (i) cover the whole imperfection shape range and (ii) can be compared in a meaningful way, the following approach was adopted:

- (i) To determine “pure” critical buckling modes with unit mid-span (i₁) mid-web flexural displacement (local-plate – $w_{LP}=I$) and (i₂) flange-lip corner vertical displacement (distortional – $v_D=I$)¹ – these two modes were obtained through preliminary linear stability analyses, based on a finite element mesh *identical* to the one adopted in the post-buckling analyses. Then, a given “combined” (critical) imperfection is obtained as a linear combination of the pure modes, with coefficients $w_{LP,0}$ and $v_{D,0}$. Note that, in general, both buckling modes will contribute to w_0 and v_0 , *i.e.*, one has

$$w_0 = w_{D,0} + w_{LP,0} \quad v_0 = v_{D,0} + v_{LP,0} \quad (1)$$

where $w_{D,0}$, $w_{LP,0}$, $v_{D,0}$ and $v_{LP,0}$ quantify the aforementioned contributions.

- (ii) All initial imperfections share the same overall magnitude, equal to 10% of the wall thickness t . In order to achieve this, one begins by normalising the pure modes in such a way that $w_{LP,0}=0.1 t$ and $v_{D,0}=0.1 t$ (in this particular case, $0.1 t=0.1 \text{ mm}$). Then, one ensures the above combined amplitude by simply enforcing the condition

$$(v_{D,0})^2 + (w_{LP,0})^2 = 0.1^2 \quad (2)$$

¹ In order to be able to “separate” the local-plate and distortional modes, it was necessary to perform buckling FEA in columns with slightly altered wall thickness values.

- (iii) A better visualisation and “feel” of the initial imperfection shape can be obtained by looking at the 0.1 mm radius circle drawn in the $w_{LP,0}-v_{D,0}$ plane and shown in Figure. 2(a): each “acceptable” imperfection shape lies on this circle and corresponds to an angle θ , measured counter clockwise from the horizontal ($v_{D,0}$) axis and defining the ratio $v_{D,0} / w_{LP,0}$ ($v_{D,0}=0.1\cos\theta$ and $w_{LP,0}=0.1\sin\theta$). Figure. 2(b) shows the FEM-based initial imperfections associated with $\theta=0, 180^\circ$ and $\theta=90, 270^\circ$ – pure distortional and pure local-plate. Finally, note that (iii₁) $\theta=0^\circ$ and $\theta=180^\circ$ correspond to *inward* and *outward* flange-lip motions and (iii₂) $\theta=90^\circ$ and $\theta=270^\circ$ to *outward* and *inward* mid-span web bending.
- (iv) In this work, initial imperfections associated with angles multiple of 15° were considered – *i.e.*, the 14 imperfection shapes defined by $\theta=0, 30, 45, 60, 90, 120, 150, 180, 210, 240, 270, 300, 315, 330^\circ$.

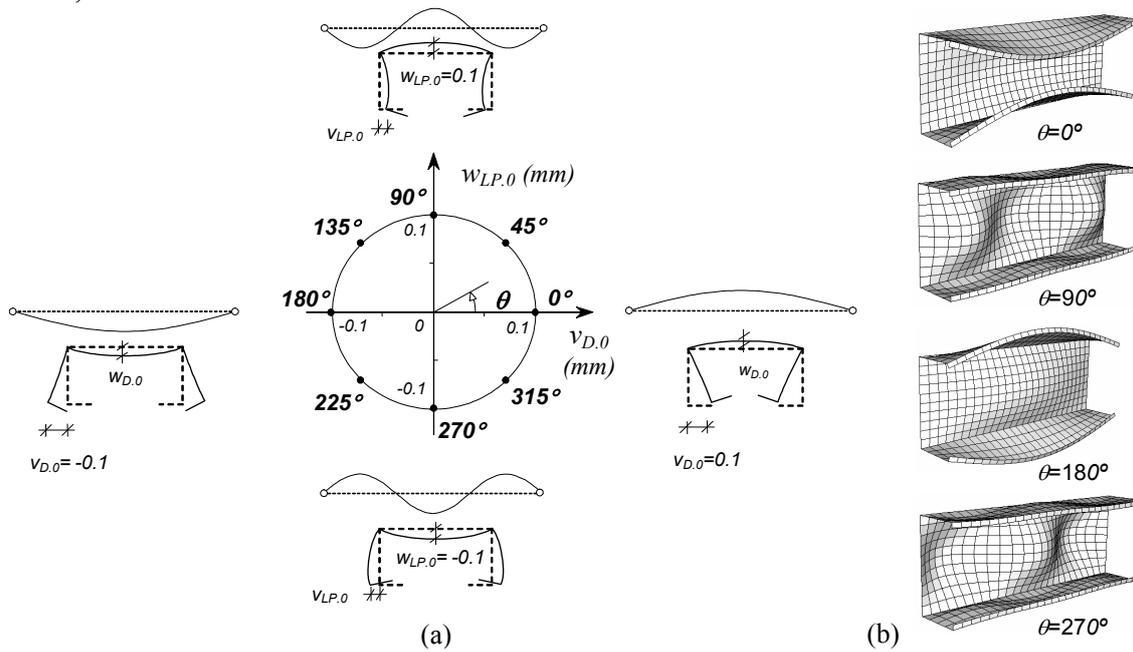


Figure 2. (a) Initial Geometrical Imperfection Representation in the $w_{LP,0}-v_{D,0}$ Plane and (b) Four FEM-Based Imperfection Shapes ($\theta=0, 90, 180, 270^\circ$)

2.2 Post-Buckling Equilibrium Paths

One addresses now the influence of the local-plate/distortional mode interaction in the elastic-plastic post-buckling behaviour of lipped channel columns (i) containing the 14 initial imperfection shapes defined above (all with the same overall amplitude) and (ii) exhibiting the 5 yield-to-critical stress ratios f_y/σ_{cr} given earlier (recall that one has $\sigma_{cr,LP}=\sigma_{cr,D}=100.5 \text{ MPa}$, which corresponds to $P_{cr}=21.1 \text{ kN}$). Figures. 3(a)-(b) show the upper portions ($P/P_{cr} > 0.8$) of the elastic and elastic-plastic equilibrium paths P/P_{cr} vs. v/t describing the post-buckling behaviours of columns with (i) initial imperfections defined by $0 \leq \theta \leq 180^\circ$ (Figure. 3(a)) and $180 \leq \theta \leq 360^\circ$ (Figure. 3(b)), and (ii) yield-to-critical stress ratios $f_y/\sigma_{cr} \approx 2, 3.5, 5.5$. On the other hand, Figures. 4(a)-(b) show the upper portions ($P/P_{cr} > 0.7$) of similar equilibrium paths of columns with the lowest stress ratio ($f_y/\sigma_{cr} \approx 1.2$). Finally, Figures. 5(a)-(b) provide information about the evolution of the plastic strains in the two columns defined by $\theta=0^\circ + f_y/\sigma_{cr} \approx 3.5$ and $\theta=90^\circ + f_y/\sigma_{cr} \approx 3.5$ – in each case, one presents 4 plastic strain diagrams, corresponding to different equilibrium states located along the post-buckling equilibrium path (their locations are indicated in Figure. 5(a)). It is worth noting that (i) the deformed configurations corresponding to the points **1** are amplified 10 times with respect to the remaining ones and that (ii) the points **4** always correspond to equilibrium states immediately after the column collapse – *i.e.*, the associated deformed configurations provide information about the column *failure modes*.

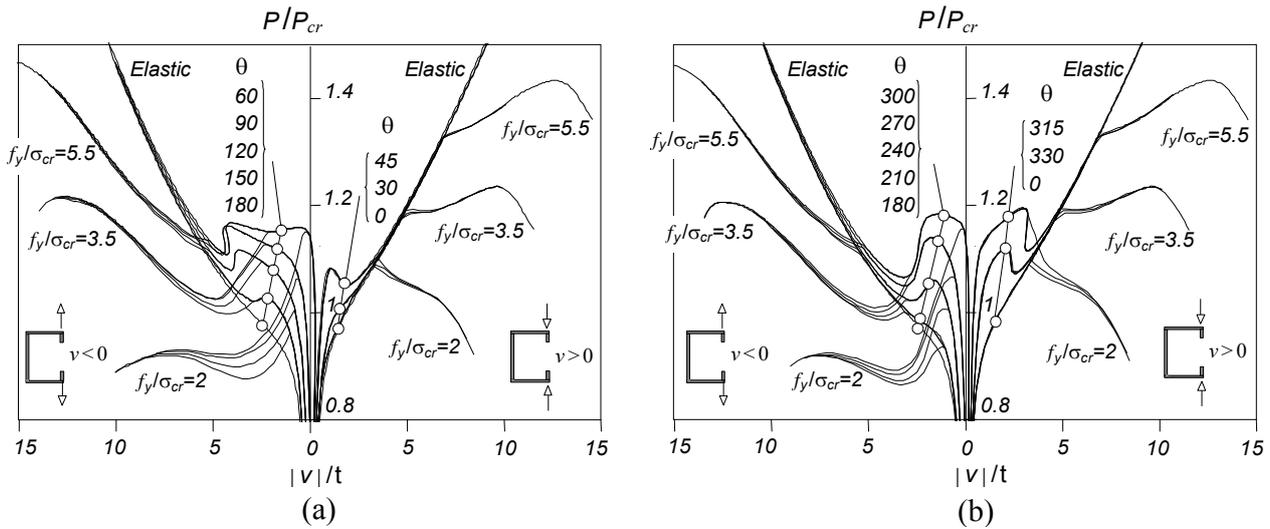


Figure 3. Column Elastic and Elastic-Plastic ($f_y/\sigma_{cr} \approx 2, 3.5, 5.5$) Post-Buckling Equilibrium Paths P/P_{cr} vs. v/t : (a) $0^\circ \leq \theta \leq 180^\circ$ and (b) $180^\circ \leq \theta \leq 360^\circ$

The observation of the post-buckling results presented in Figures. 3 to 5 prompts the following remarks:

- (i) Even under local-plate/distortional (LP/D) mode interaction, there is a visible elastic post-buckling asymmetry concerning the columns with pure distortional imperfections: the $\theta=0^\circ$ (inward) column exhibits slightly higher post-buckling strengths than its $\theta=180^\circ$ (outward) counterpart. However, note that, regardless of the imperfection “sign”, the column post-buckling strength is always a bit higher than in the absence of LP/D interaction, which stems from the progressive emergence of a (rather small) local-plate component, known to exhibit a much larger post-critical strength reserve – Figures 6(a)-(b) compare the evolutions of the $\theta=0^\circ$ and $\theta=180^\circ$ column web deformed configurations with an *exact sinusoid shape* – these comparisons clearly reveal the presence of the local-plate 3 half-wave sinusoidal component, thus confirming the occurrence of the buckling mode interaction [16].
- (ii) All elastic equilibrium paths associated with predominantly local-plate imperfections (namely the ones concerning the $\theta=90^\circ$ and $\theta=270^\circ$ columns) exhibit a distinct “irregular” behaviour: regardless of the w_{LP0} sign, they (ii₁) are less “smooth” than the $\theta=0^\circ$ and $\theta=180^\circ$ column paths, (ii₂) exhibit limit points, associated with “snap-through” phenomena², and, for $P/P_{cr} > 1.1$, (ii₃) merge with either the $\theta=180^\circ$ column (majority of them) or $\theta=0^\circ$ column paths. This means that, even in the presence of pure 3 half-wave local-plate imperfections ($v_{D,0}=0$), the column evolves towards a predominantly distortional single half-wave deformed configuration – this evolution often includes significant web *bending reversal*, which explains the occurrence of the limit points appearing in the (elastic) equilibrium paths displayed in Figures. 3(a)-(b).
- (iii) Generally speaking, the elastic-plastic equilibrium paths of the various columns also merge with the ones corresponding to $\theta=0^\circ$ or $\theta=180^\circ$. Depending on the f_y/σ_{cr} value, this merging may occur after or before the column has reached its ultimate strength – see Figures. 3(a)-(b). However, merging never occurs for $f_y/\sigma_{cr} \approx 1.2$ – an explanation for this fact is provided below, in item (vi).

² The “snap-through” becomes more “abrupt” as the amplitude of the initial imperfection local-plate component grows. When the imperfections are pure or “almost pure”, one even observes a minor “snap-back” phenomenon (e.g., the $\theta=90^\circ$ column).

- (iv) The onset of yielding always takes place inside the load interval $0.8 < P/P_{cr} < 1.3$ (obviously, the exact load value depends on the yield-to-critical stress ratio and initial imperfection shape) and defines the point of separation between the elastic and elastic-plastic equilibrium paths. For a given f_y/σ_{cr} value, this separation takes place for higher load levels in columns with pure local-plate imperfections ($\theta=90^\circ$ or $\theta=270^\circ$) than in columns with pure distortional ones ($\theta=0^\circ$ or $\theta=180^\circ$) – the difference may reach about 20% (e.g., $\theta=180^\circ$ vs. $\theta=270^\circ$, for $f_y/\sigma_{cr}\approx 2$).
- (v) For $f_y/\sigma_{cr}\geq 2$, plasticity appears first at the free ends of the mid-span cross-section section lips, as shown in Figures. 5(b₁)-(b₂).
- (vi) For $f_y/\sigma_{cr}\approx 1.2$, yielding starts while the column is still subjected to a fairly uniform stress distribution, leading to a very “sudden” collapse – a large portion of the column yields at practically the same time. In this case, the shape of the initial imperfection governs the location of zone where yielding begins.
- (vii) For $f_y/\sigma_{cr}\geq 3.5$, collapse only occurs after the full yielding of the column mid-span zones located around the web-flange corners, as shown in the right side of Figures. 5(b₁)-(b₂). However, one should point out that the collapse mechanism is different in columns with outward and inward flange-lip motions. In the former, one observes the formation of a “three hinge flange mechanism” – see the right side of Figure. 5(b₂). In the latter, the plastic deformation is restricted to the close vicinity of the mid-span cross-section – see the right side of Figure. 5(b₁).
- (viii) For $f_y/\sigma_{cr}\geq 3.5$, the onset of yielding does not precipitate failure, even when associated with a “snap-through” phenomenon (e.g., $f_y/\sigma_{cr}\approx 3.5$ and any imperfection shape with a local-plate component) – the column still exhibits a certain amount of post-buckling strength reserve, which is more substantial for $f_y/\sigma_{cr}\approx 5.5$ and outward flange-lip motion – see Figure. 3(a). It is still worth noting that failure occurs after the various equilibrium paths have already merged together, which means that the initial imperfection shape has little influence on the column load-carrying capacity and failure mechanism.
- (ix) The outward pure distortional initial imperfection is the most *detrimental* one, as it leads to the lowest column load-carrying capacities. This feature has far-reaching implications in the design of cold-formed steel columns affected by local-plate/distortional mode interaction – see the next subsection.

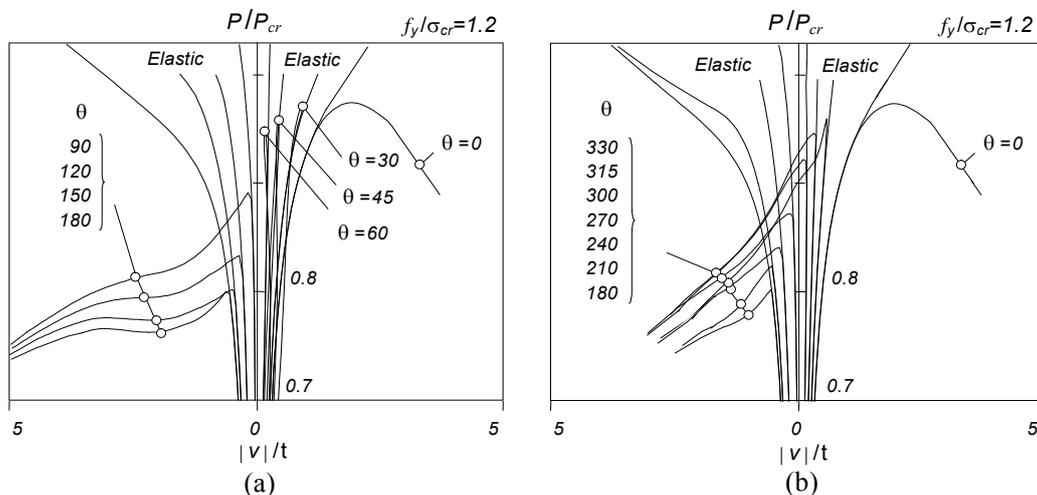


Figure 4. Column Elastic and Elastic-Plastic ($f_y/\sigma_{cr}\approx 1.2$) Post-Buckling Equilibrium Paths P/P_{cr} vs. v/t : (a) $0^\circ \leq \theta \leq 180^\circ$ and (b) $180^\circ \leq \theta \leq 360^\circ$

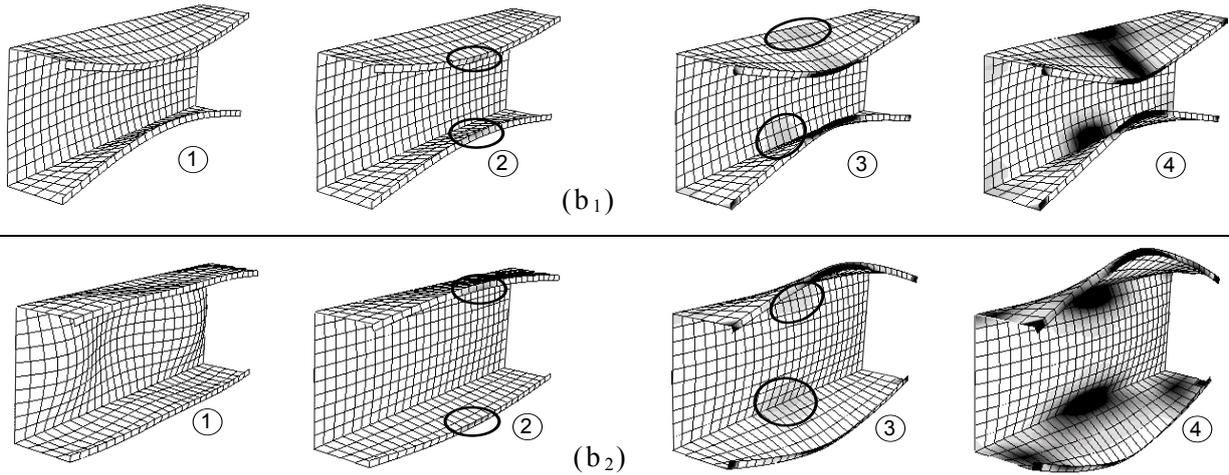
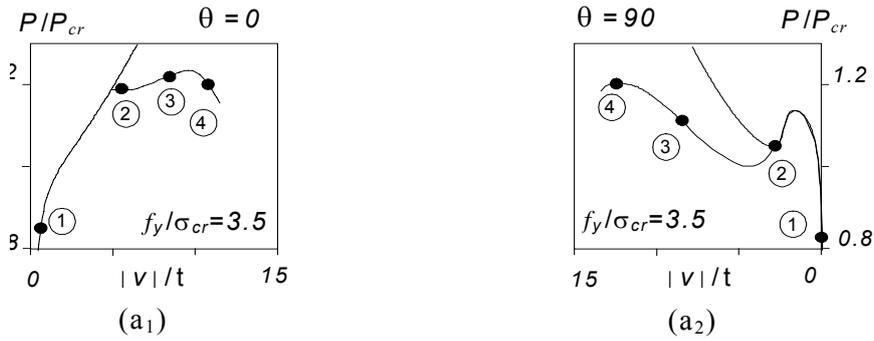


Figure 5. (a) Equilibrium Points on P/P_{cr} vs. v/t Post-Buckling Paths and (b) Associated Plastic Strain Diagrams and Failure Modes for Columns Defined by: (1) $\theta=0^\circ + f_y/\sigma_{cr}\approx 3.5$ and (2) $\theta=90^\circ + f_y/\sigma_{cr}\approx 3.5$

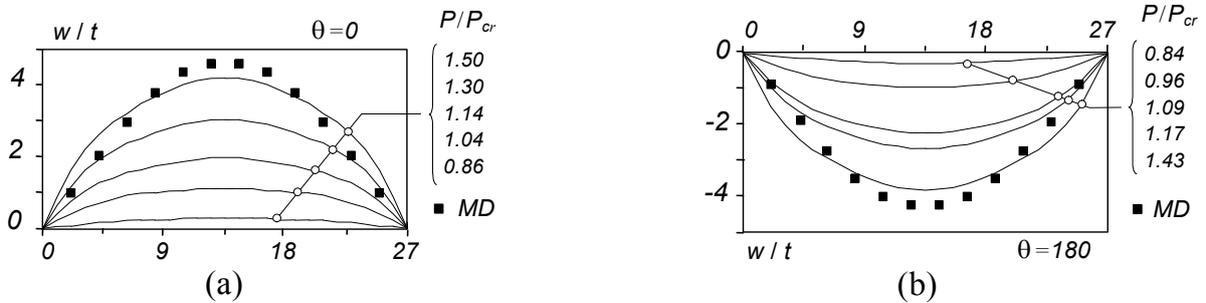


Figure 6. Web Deformed Configuration Evolution for the (a) $\theta=0^\circ$ and (b) $\theta=180^\circ$ Columns

3. DIRECT STRENGTH METHOD

Next, one presents the results of an investigation aimed at developing and assessing the performance of an approach based on the Direct Strength Method (DSM) and intended to estimate the ultimate strength of lipped channel columns affected by LP/D mode interaction [12, 13]. In order to achieve this goal in a proper fashion, it is indispensable to possess a significant amount of reliable experimental and/or numerical ultimate strength values concerning columns with close (almost coincident) local-plate and distortional critical buckling stresses. Therefore, it was decided to carry out an extensive parametric study, destined to acquire a fairly large data bank of column ultimate strengths on which to base the assessment and/or improvement of the current DSM design methodology – these ultimate strengths were obtained through

second-order elastic-plastic shell finite element analyses performed in the code ABAQUS [17], as briefly described earlier.

3.1 Current DSM Design Approach

When compared with the traditional “*effective width* approach”, the DSM exhibits three major innovative features, all stemming from the fact that the cross-section is now viewed as a *whole*: (i) wall-restraint effects are automatically taken into account, (ii) no effective width calculations are needed and (iii) it is possible to provide strength estimates for members failing in *distortional* modes. Moreover, the DSM provides a rational and systematic framework for the design of thin-walled members with arbitrary cross-section shapes, loadings or failure modes – of course, a given application needs proper calibration and validation (*i.e.*, comparison with a fair number of experimental and/or numerical results). Finally, note that both the DSM and effective width approaches share the same basic assumption: the member ultimate strength can be accurately predicted solely on the basis of its elastic buckling and yield stresses.

The available DSM applications adopt “Winter-type” design curves, which have been calibrated against a large number of experimental and/or numerical results [19]. It was shown that, whenever a given column fails in pure local-plate or distortional modes, it is possible to obtain safe and accurate ultimate strength estimates on the basis of the elastic buckling and yield stress values. Thus, the DSM prescribes that the column *nominal strengths* against *local-plate* and *distortional* failure (P_{NL} and P_{ND}) are given by the expressions [2]

$$\begin{cases} P_{NL} = P_Y & \text{if } \lambda_L \leq 0.776 \\ P_{NL} = P_Y \left(\frac{P_{CRL}}{P_Y} \right)^{0.4} \left[1 - 0.15 \left(\frac{P_{CRL}}{P_Y} \right)^{0.4} \right] & \text{if } \lambda_L > 0.776 \end{cases} \quad (3)$$

$$\begin{cases} P_{ND} = P_Y & \text{if } \lambda_D \leq 0.561 \\ P_{ND} = P_Y \left(\frac{P_{CRD}}{P_Y} \right)^{0.6} \left[1 - 0.25 \left(\frac{P_{CRD}}{P_Y} \right)^{0.6} \right] & \text{if } \lambda_D > 0.561 \end{cases} \quad (4)$$

where (i) $\lambda_L = (P_Y / P_{CRL})^{0.5}$ and $\lambda_D = (P_Y / P_{CRD})^{0.5}$, (ii) P_Y is the squash load and (iii) P_{CRL} and P_{CRD} are the *local-plate* and *distortional* critical buckling loads. In order to capture the local-plate/global or distortional/global interaction, the DSM approach proposes the replacement of P_Y by P_{NE} in Eq. 3 or Eq. 4, where P_{NE} is the column buckling strength associated with *global* (Euler) failure. At this point, it is worth noting that an accurate prediction of the column distortional failure load has considerable practical relevance, since (i) the distortional post-critical strength reserve is considerably lower and more imperfection-sensitive than its local-plate counterpart and (ii) there exists clear (numerical) evidence that the collapse of columns buckling in local-plate modes is often associated with a distortional failure mechanism [14].

3.2 DSM for Local-Plate/Distortional Interaction

Following a strategy similar to the one used to develop safety-checking rules that account for local-plate/global effects, it is possible to propose expressions to estimate the ultimate strength of columns experiencing local-plate/distortional interaction. This was first achieved by Schafer [20, 21], who proposed two distinct approaches: (i) replacing P_Y by P_{ND} in Eq. 3 (NLD approach – schematically depicted in the flowchart shown in Figure. 7(a)) or (ii) replacing P_Y by P_{NL} in Eq. 4 (NDL approach – see Figure. 7(b)), where P_{ND} and P_{NL} are the distortional and local-plate buckling strengths also given by Eqs. 3 and 4. Yang and Hancock [10] recently adopted the NLD approach to investigate the LP/D

interaction in lipped channel columns with “v-shape” web and flange intermediate stiffeners. After comparing the ultimate strength estimates provided by the NLD approach with the results of a series of experimental tests performed in Sydney, which provided evidence of adverse local-plate/distortional interaction, these authors concluded that (i) the above estimates were always safe and reasonably accurate (differences within the 10-20% range) and also that (ii) further investigation is required on the design of columns with nearly coincident local-plate and distortional buckling stresses. Finally, note that the use of Eqs. 3 and 4 requires the knowledge of *accurate* local-plate and distortional buckling loads (P_{CRL} , P_{CRD}), which can be obtained through shell finite element analyses, finite strip analyses or Generalised Beam Theory (GBT) analyses.

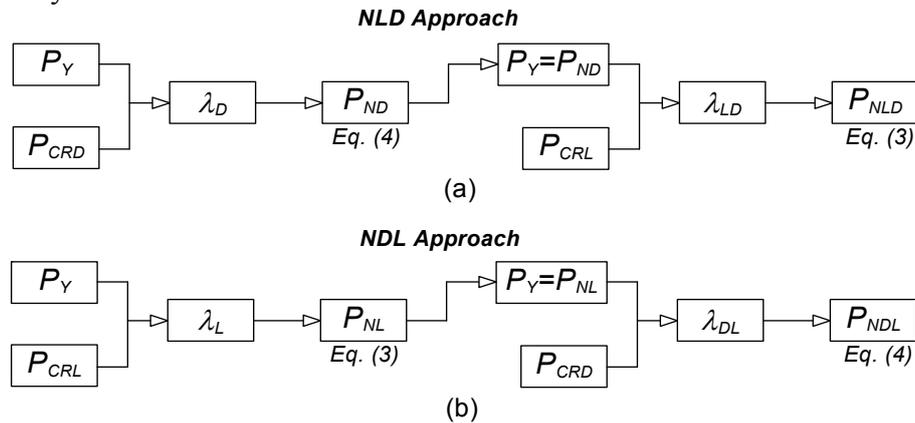


Figure 7. Flowcharts Concerning the Determination of the Nominal Strength of Columns Experiencing LP/D Interaction Effects: (a) P_{NLD} and (b) P_{NDL} Approaches

4. PARAMETRIC STUDY: SCOPE AND NUMERICAL RESULTS

In order to be able to carry out a fairly large parametric study involving the ultimate strength of lipped channel columns affected by local-plate/distortional interaction, their geometries had to be carefully selected: it was necessary to find sets of cross-section dimensions and length values making it possible to “control” the closeness between the column local-plate and distortional critical buckling stresses (σ_{CRL} and σ_{CRD}) or loads (P_{CRL} and P_{CRD}). This goal was achieved by adopting the following strategy:

- (i) Initially, a trial-and-error approach was employed to find six “basic cross-section shapes”, displaying commonly used dimensions and ensuring the coincidence between σ_{CRL} and σ_{CRD} . This search led to the three *slender* ($1.4 \leq \lambda_D \leq 2.6$) and three *stocky* ($0.6 \leq \lambda_D \leq 1.4$) cross-sections exhibiting the following dimensions (web width b_w , flange width b_f , stiffener/lip width b_s and wall thickness t)
 - (i.1) Slender cross-sections: (i₁) $b_w=100$, $b_f=50$, $b_s=5$, $t=1.0mm$ (Slender Column 1 – SLC1), (i₂) $b_w=120$, $b_f=80$, $b_s=10$, $t=1.3mm$ (SLC2) and (i₃) $b_w=95$, $b_f=80$, $b_s=10$, $t=0.95mm$ (SLC3).
 - (i.2) Stocky cross-sections: (i₁) $b_w=180$, $b_f=100$, $b_s=20$, $t=3.4mm$ (Stocky Column 1 – STC1), (i₂) $b_w=110$, $b_f=78$, $b_s=30$, $t=2.8mm$ (STC2) and (i₃) $b_w=100$, $b_f=100$, $b_s=26$, $t=2.0mm$ (STC3).
 One should mention that all the above six column cross-section dimensions satisfy the requirements that have been adopted in the existing DSM approach – i.e., they are “pre-qualified columns”.
- (ii) Subsequently, the closeness between σ_{CRL} and σ_{CRD} was slightly altered, by just changing a single basic cross-section dimension: either the flange width b_f , the web width b_w or the stiffener width b_s . This procedure led to the identification of various columns with (ii₁) cross-section dimensions generated from the six basic shapes and (ii₂) very close (but not necessarily coincident) σ_{CRL} and σ_{CRD} values – the alterations were made in such a way that one has always $0.90 \leq \sigma_{CRL} / \sigma_{CRD} \leq 1.10$ (i.e., the critical stresses are never more than 10% apart).

Concerning the column lengths considered, they always correspond to single distortional half-waves associated with the buckling stresses σ_{CRD} and were determined by means of finite strip analyses. As for the column steel material behaviour, it is characterised by $E=210 \text{ GPa}$ (Young's modulus), $\nu=0.3$ (Poisson's ratio) and $f_y=250\text{-}350\text{-}550 \text{ MPa}$ (three yield stress values are considered in this work, all of them also meeting the DSM limit stress requirements for “pre-qualified columns”). Finally, it is worth (i) mentioning that no residual stresses have been taken into account (it has been shown that have very little impact on the column ultimate strength – e.g., [22, 23]) and (ii) addressing the criterion adopted to select the initial geometrical imperfections included in the non-linear analyses that provide the column ultimate strengths:

- (i) Regardless of their critical stress ratios $\sigma_{CRL} / \sigma_{CRD}$, all the columns analysed contained initial geometrical imperfections with a single-wave distortional buckling mode shape, having an amplitude (mid-span flange-lip corner displacement) equal to 10% of the wall thickness t and involving *outward* motions of the flange-lip assemblies – previous studies involving lipped channel columns with $\sigma_{CRL} = \sigma_{CRD}$ showed that this imperfection shape is the *most detrimental* one, in the sense that it corresponds to the lowest column post-buckling strength and collapse load [15, 16].
- (ii) The *slender* columns with $\sigma_{CRL} / \sigma_{CRD} < 1.0$ (i.e., with critical *local-plate* buckling modes exhibiting several half-waves) were also analysed in the presence of *critical-mode* initial geometrical imperfections, again with amplitude $0.1 t$ – it now corresponds to the mid-web flexural displacement at mid-span.
- (iii) All initial geometrical imperfections defined earlier (buckling mode shapes with amplitude $0.1 t$) are included in the analyses through a specific ABAQUS command. In the columns buckling in local-plate modes ($\sigma_{CRL} < \sigma_{CRD}$), the single-wave distortional imperfection is, effectively, the column higher-order buckling mode most resembling it, which means that it is not possible to guarantee the complete “purity” of the distortional shape – in other words, a small participation of a multiple half-wave local-plate mode is virtually undetectable (but with only a very minute affect the column ultimate strength).

A total of 66 *slender* and 45 *stocky* columns were analysed, corresponding to all possible combinations of 16 (15) different cross-section shapes and 3 yield stress values. All cross-section dimensions (b_w, b_f, b_s, t), lengths (L), critical stresses ($\sigma_{CRL}, \sigma_{CRD}$), yield stresses (f_y) and initial imperfection shapes (D, LP) considered in this work are given in Tables 1-2 (*slender* columns) and 3-4 (*stocky* columns), presented further ahead.

The numerical results displayed in Tables 1 to 4 consist of column average stresses at collapse ($\sigma_U = P_U / A$) and the nature of the corresponding failure mechanisms. In order to convey the meaning of these results, they are illustrated in Figure. 8(a), which shows the post-buckling equilibrium paths (σ / σ_{CR} vs. v/t) concerning columns with (i) $\sigma_{CRL} = \sigma_{CRD}$ ($\equiv \sigma_{CR}$), (ii) identical outward distortional imperfections and (iii) four different yield stresses ($f_y / \sigma_{CR} \approx 1.2, 2.0, 3.5, 5.5$). It is worth noting that the onset of yielding, which always takes place in the stiffener free ends (see Figure. 8(b₁)), occurs at the equilibrium points **A** and may or may not trigger the column failure – it depends on the particular f_y / σ_{CR} value. Indeed, for large enough f_y / σ_{CR} values, failure occurs at a limit point **B**, following (i) a “snap-through” phenomenon and (ii) the yielding of the column central regions located around the web-flange corners (see Figure. 8(b₂)) – Figure. 8(c) shows the corresponding (predominantly distortional) failure mechanism.

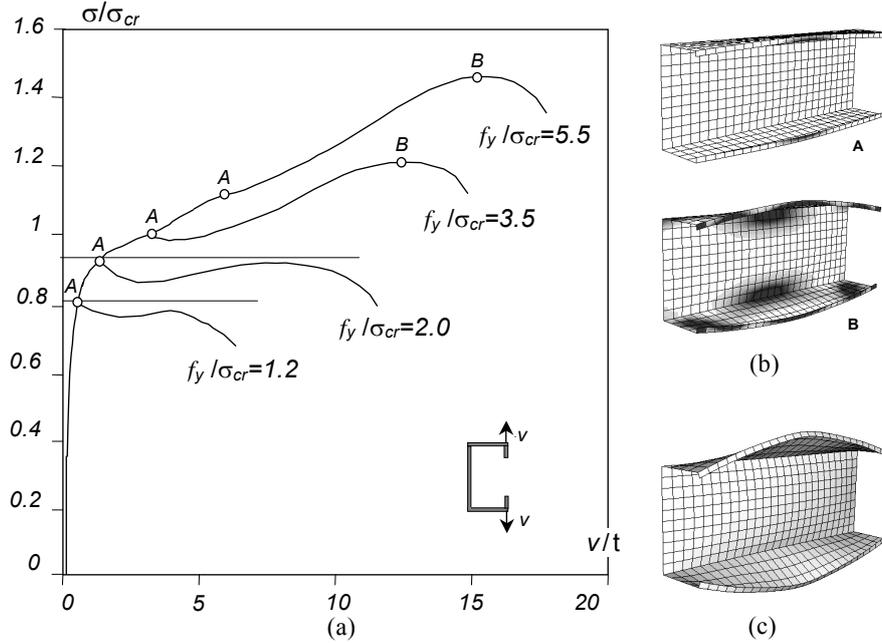


Figure 8. Column (a) Post-Buckling Equilibrium Paths, (b) Plastic Strain Distributions and (c) Failure Mechanism

Table 1(a). Comparison Between the “Exact” and DSM (σ_{NLD}) Ultimate Strength Estimates (SLC1)

	b_f	L	Imp.	f_y	FEA				DSM				
					σ_L	σ_D	σ_U	Fail.	λ_D	σ_{ND}	λ_{LD}	σ_{NLD}	σ_{NLD}/σ_U
$b_w=100\text{mm}, b_s=5\text{mm}, t=1\text{mm}$	55	300	D	250	101	91	94	B	1.66	117	1.08	95	1.01
				350			110	B	1.97	138	1.17	106	0.96
				550			131	A	2.46	171	1.30	121	0.93
	52.5	280	D	250	101	96	97	B	1.61	121	1.09	97	1.00
				350			114	B	1.91	143	1.19	108	0.95
				550			137	A	2.39	176	1.32	124	0.91
	50	270	D	250	102	102	102	B	1.57	125	1.11	99	0.97
				350			120	B	1.85	147	1.20	110	0.92
				550			147	A	2.32	182	1.34	127	0.86
	47.5	260	D	250	102	108	107	B	1.52	128	1.12	101	0.94
				350			127	B	1.80	151	1.22	113	0.89
				550			156	A	2.26	187	1.35	130	0.83
	45	260	D	250	103	113	115	B	1.48	131	1.13	103	0.90
				350			136	B	1.76	155	1.23	115	0.85
				550			168	A	2.20	193	1.37	132	0.79
	47.5	260	LP	250	102	108	118	A	1.52	128	1.12	101	0.86
				350			127	B	1.80	151	1.22	113	0.89
				550			157	B	2.26	187	1.35	130	0.83
	45	260	LP	250	103	113	128	A	1.48	131	1.13	103	0.80
				350			142	B	1.76	155	1.23	115	0.81
				550			168	B	2.20	193	1.37	132	0.79

Table 1(b). Comparison Between the “Exact” and DSM (σ_{NLD}) Ultimate Strength Estimates (SLC2)

	b_w	L	Imp.	f_y	FEA				DSM				
					σ_L	σ_D	σ_U	Fail.	λ_D	σ_{ND}	λ_{LD}	σ_{NLD}	σ_{NLD}/σ_U
$b_f=80\text{mm}, b_s=10\text{mm}, t=1.3\text{mm}$	130	550	D	250	100	110	105	A	1.51	129	1.14	101	0.96
				350			107	B	1.78	153	1.24	113	1.05
				550			123	B	2.24	189	1.38	130	1.05
	125	550	D	250	107	113	107	A	1.49	131	1.11	104	0.97
				350			109	A	1.76	155	1.20	116	1.07
				550			123	B	2.21	192	1.34	134	1.09
	120	550	D	250	115	115	109	A	1.47	133	1.07	108	0.99
				350			111	A	1.74	157	1.17	120	1.08
				550			124	B	2.18	194	1.30	139	1.12
	115	550	D	250	124	118	112	A	1.45	134	1.04	111	0.99
				350			114	A	1.72	159	1.13	124	1.09
				550			122	B	2.16	197	1.26	143	1.18
	110	550	D	250	135	121	114	A	1.44	136	1.00	115	1.01
				350			116	A	1.70	161	1.09	129	1.11
				550			121	A	2.13	199	1.22	149	1.23
	100	550	D	250	157	127	119	A	1.40	139	0.94	123	1.03
				350			122	A	1.66	164	1.02	138	1.13
				550			126	A	2.08	204	1.14	159	1.26
	125	550	LP	250	107	113	119	A	1.49	131	1.11	104	0.88
				350			120	A	1.76	155	1.20	116	0.97
				550			122	B	2.21	192	1.34	134	1.10
	130	550	LP	250	100	110	120	A	1.51	129	1.14	101	0.84
				350			121	A	1.78	153	1.24	113	0.93
				550			123	B	2.24	189	1.38	130	1.05

Table 1(c). Comparison Between the “Exact” and DSM (σ_{NLD}) Ultimate Strength Estimates (SLC3)

	b_s	L	Imp.	f_y	FEA				DSM				
					σ_L	σ_D	σ_U	Fail.	λ_D	σ_{ND}	λ_{LD}	σ_{NLD}	σ_{NLD}/σ_U
$b_w=95\text{mm}, b_f=80\text{mm}, t=0.95\text{mm}$	11	650	D	250	92	100	94	A	1.58	123	1.16	95	1.01
				350			95	A	1.87	145	1.26	106	1.11
				550			99	A	2.35	180	1.40	122	1.23
	10.5	650	D	250	92	96	91	A	1.62	121	1.15	94	1.03
				350			92	A	1.91	142	1.25	104	1.13
				550			97	A	2.40	176	1.39	120	1.24
	10	600	D	250	91	91	86	A	1.65	118	1.14	92	1.07
				350			87	A	1.96	139	1.23	103	1.17
				550			92	A	2.45	171	1.37	118	1.28
	9.5	600	D	250	91	87	83	A	1.70	115	1.12	91	1.10
				350			84	A	2.01	135	1.22	101	1.20
				550			91	B	2.51	167	1.35	116	1.27
	9	550	D	250	91	83	78	A	1.74	112	1.11	89	1.14
				350			79	A	2.06	132	1.20	99	1.25
				550			84	A	2.58	162	1.34	114	1.35
	10.5	650	LP	250	92	96	109	A	1.62	121	1.15	94	0.86
				350			109	A	1.91	142	1.25	104	0.96
				550			109	A	2.40	176	1.39	120	1.10
	11	650	LP	250	92	100	114	A	1.58	123	1.16	95	0.83
				350			114	A	1.87	145	1.26	106	0.93
				550			114	A	2.35	180	1.40	122	1.07

Table 2(a). Comparison Between the “Exact” and DSM (σ_{NDL}) Ultimate Strength Estimates (SLC1)

	b_f	L	Imp.	f_y	FEA				DSM				σ_{NDL}/σ_U
					σ_L	σ_D	σ_U	Fail.	λ_L	σ_{NL}	λ_{DL}	σ_{NDL}	
$b_w=100\text{mm}, b_s=5\text{mm}, t=1\text{mm}$	55	300	D	250	101	91	94	B	1.58	156	1.31	92	0.98
				350			110	B	1.86	193	1.46	103	0.94
				550			131	A	2.34	258	1.69	119	0.91
	52.5	280	D	250	101	96	97	B	1.57	156	1.27	95	0.98
				350			114	B	1.86	194	1.42	106	0.93
				550			137	A	2.33	258	1.64	123	0.90
	50	270	D	250	102	102	102	B	1.57	156	1.24	97	0.96
				350			120	B	1.85	194	1.38	109	0.91
				550			147	A	2.32	259	1.59	127	0.86
	47.5	260	D	250	102	108	107	B	1.56	157	1.21	100	0.94
				350			127	B	1.85	194	1.34	112	0.89
				550			156	A	2.32	259	1.55	130	0.84
	45	260	D	250	103	113	115	B	1.56	157	1.18	103	0.89
				350			136	B	1.84	195	1.31	115	0.85
				550			168	A	2.31	260	1.51	134	0.80
	47.5	260	LP	250	102	108	118	A	1.56	157	1.21	100	0.85
				350			127	B	1.85	194	1.34	112	0.89
				550			157	B	2.32	259	1.55	130	0.83
45	260	LP	250	103	113	128	A	1.56	157	1.18	103	0.80	
			350			142	B	1.84	195	1.31	115	0.81	
			550			168	B	2.31	260	1.51	134	0.80	

Table 2(b). Comparison Between the “Exact” and DSM (σ_{NDL}) Ultimate Strength Estimates (SLC2)

	b_w	L	Imp.	f_y	FEA				DSM				σ_{NDL}/σ_U
					σ_L	σ_D	σ_U	Fail.	λ_L	σ_{NL}	λ_{DL}	σ_{NDL}	
$b_f=80\text{mm}, b_s=10\text{mm}, t=1.3\text{mm}$	130	550	D	250	100	110	105	A	1.58	155	1.19	101	0.96
				350			107	B	1.87	193	1.32	113	1.06
				550			123	B	2.35	257	1.53	131	1.07
	125	550	D	250	107	113	107	A	1.53	159	1.19	103	0.96
				350			109	A	1.81	198	1.32	116	1.06
				550			123	B	2.27	264	1.53	135	1.09
	120	550	D	250	115	115	109	A	1.47	163	1.19	106	0.97
				350			111	A	1.74	203	1.33	119	1.07
				550			124	B	2.18	271	1.53	138	1.11
	115	550	D	250	124	118	112	A	1.42	168	1.19	108	0.97
				350			114	A	1.68	209	1.33	122	1.07
				550			122	B	2.10	278	1.53	142	1.16
	110	550	D	250	135	121	114	A	1.36	172	1.19	111	0.98
				350			116	A	1.61	214	1.33	125	1.08
				550			121	A	2.02	287	1.54	145	1.20
	100	550	D	250	157	127	119	A	1.26	182	1.20	117	0.98
				350			122	A	1.49	226	1.34	132	1.08
				550			126	A	1.87	303	1.55	153	1.21
125	550	LP	250	107	113	119	A	1.53	159	1.19	103	0.87	
			350			120	A	1.81	198	1.32	116	0.97	
			550			122	B	2.27	264	1.53	135	1.10	
130	550	LP	250	100	110	120	A	1.58	155	1.19	101	0.84	
			350			121	A	1.87	193	1.32	113	0.93	
			550			123	B	2.35	257	1.53	131	1.07	

Table 2(c). Comparison Between the “Exact” and DSM (σ_{NDL}) Ultimate Strength Estimates (SLC3)

	b_s	L	Imp.	f_y	FEA				DSM				
					σ_L	σ_D	σ_U	Fail.	λ_L	σ_{NL}	λ_{DL}	σ_{NDL}	σ_{NDL}/σ_U
b _w =95 mm, b _f =80 mm, t=0.95 mm	11	650	D	250	92	100	94	A	1.65	150	1.23	95	1.00
				350			95	A	1.96	187	1.37	106	1.12
				550			99	A	2.45	249	1.58	123	1.24
	10.5	650	D	250	92	96	91	A	1.65	150	1.25	93	1.02
				350			92	A	1.96	187	1.40	104	1.13
				550			97	A	2.45	249	1.61	120	1.24
	10	600	D	250	91	91	86	A	1.65	150	1.28	91	1.05
				350			87	A	1.96	187	1.43	102	1.17
				550			92	A	2.45	249	1.65	118	1.28
	9.5	600	D	250	91	87	83	A	1.66	150	1.31	89	1.08
				350			84	A	1.96	187	1.46	99	1.19
				550			91	B	2.46	248	1.69	115	1.26
	9	550	D	250	91	83	78	A	1.66	150	1.35	87	1.11
				350			79	A	1.96	186	1.50	97	1.22
				550			84	A	2.46	248	1.73	112	1.33
	10.5	650	LP	250	92	96	109	A	1.65	150	1.25	93	0.85
				350			109	A	1.96	187	1.40	104	0.95
				550			109	A	2.45	249	1.61	120	1.10
	11	650	LP	250	92	100	114	A	1.65	150	1.23	95	0.83
				350			114	A	1.96	187	1.37	106	0.93
				550			114	A	2.45	249	1.58	123	1.08

5. ASSESSMENT OF THE DSM ESTIMATES

The numerical and DSM results given in Tables 1 and 2 concern the 66 slender columns analysed (geometries SLC1, SLC2 and SLC3) and enable the comparison between the “exact” ultimate strengths ($\sigma_U = P_U / A$) and the estimates yielded by the two proposed DSM approaches, namely $\sigma_{NDL} = P_{NDL} / A$ and $\sigma_{NLD} = P_{NLD} / A$ – also indicated are the values of the σ_{ND} and σ_{NL} estimates. The observation of the results presented in these two tables prompts the following remarks:

- (i) The σ_U values concerning local-plate (LP) imperfections never fall below their distortional (D) counterparts, thus confirming that the distortional imperfections are the *most detrimental* ones. As the DSM does not capture the effect of the imperfection shape, its estimates should preferably approximate well the σ_U values concerning the D imperfections. If this is the case, then the DSM estimates will be more or less conservative for columns containing LP imperfection components.
- (ii) The σ_{NDL} and σ_{NLD} estimates are very similar, even if the “quality” of the former is slightly higher – the σ_{NDL} / σ_U and σ_{NLD} / σ_U means and standard deviations are 1.01 vs. 1.02 and 0.13 vs. 0.14. Although the whole sets of σ_{NDL} and σ_{NLD} estimates may be viewed as quite accurate (means very close to 1.00), one must also recognise that their scatters are rather high (standard deviations above 0.12).
- (iii) Out of the 66 σ_{NDL} estimates, 1 is exact, 19 are safe and accurate ($\sigma_{NDL} / \sigma_U \geq 0.9$), 16 are too safe ($0.80 \leq \sigma_{NDL} / \sigma_U < 0.90$), 15 are a bit unsafe ($\sigma_{NDL} / \sigma_U \leq 1.10$) and 15 are too unsafe ($1.10 < \sigma_{NDL} / \sigma_U \leq 1.33$).

- (iv) Out of the 66 σ_{NLD} estimates, 1 is exact, 16 are safe and accurate ($\sigma_{NLD} / \sigma_U \geq 0.9$), 15 are too safe ($0.79 \leq \sigma_{NLD} / \sigma_U < 0.90$), 17 are a bit unsafe ($\sigma_{NLD} / \sigma_U \leq 1.10$) and 17 are too unsafe ($1.10 < \sigma_{NLD} / \sigma_U \leq 1.35$).
- (v) Both the local-plate (σ_{NL}) and distortional (σ_{ND}) DSM expressions clearly overestimate the ultimate strengths (σ_U) of the slender columns affected by local-plate/distortional mode interaction.

As for Tables 3 and 4, they display similar numerical ($\sigma_U = P_U / A$) and DSM ($\sigma_{NDL} = P_{NDL} / A$, $\sigma_{NLD} = P_{NLD} / A$) results for the 45 stocky columns considered in this study (geometries STC1, STC2 and STC3). The observation of these results leads to the following comments:

- (i) Although the σ_{NDL} and σ_{NLD} estimates are once more very similar, the slightly higher “quality” belongs this time to the latter – the σ_{NDL} / σ_U and σ_{NLD} / σ_U means and standard deviations are 0.88 vs. 0.90 and 0.061 vs. 0.057. One notices that the whole sets of σ_{NDL} and σ_{NLD} estimates are now a bit conservative (means around 0.90), but exhibit relatively low scatters (standard deviations around 0.06) – moreover, there are no unsafe predictions at all.
- (ii) Out of the 45 σ_{NLD} estimates, 24 are both safe and accurate ($\sigma_{NLD} / \sigma_U \geq 0.9$) and 21 are excessively safe ($0.80 \leq \sigma_{NLD} / \sigma_U < 0.90$).
- (iii) Out of the 45 σ_{NDL} estimates, 19 are both safe and accurate ($\sigma_{NDL} / \sigma_U \geq 0.9$) and 26 are excessively unsafe ($0.78 \leq \sigma_{NDL} / \sigma_U < 0.90$).
- (iv) Unlike in the case of the slender columns, the distortional DSM expressions (σ_{ND}) now provide reasonably accurate ultimate strength estimates for the stocky columns affected by local-plate/distortional interaction – indeed, only in 8 columns (out of 45), all of them concerning the higher yield stress ($f_y = 550 \text{ MPa}$), are the σ_{ND} estimates larger than the σ_U values. Conversely, the local-plate DSM expressions (σ_{NL}) often lead mostly to an overestimation of the stocky column σ_U values – only in 10 columns do the σ_{NL} values fall below the σ_U ones.

In view of the facts outlined and explained above, the authors believe that the σ_{NLD} approach is superior to the σ_{NDL} one (although the differences are not at all substantial) – therefore, the results presented hereafter are based on the assumption that the σ_{NLD} approach is adopted. In this context, Figures. 9 (slender columns) and 10 (stocky columns) make it possible to assess how the stress ratios $\sigma_{CRD} / \sigma_{CRL}$ and σ_{NLD} / σ_U vary with the cross-section dimensions b_f , b_w , b_s – the latter for $f_y = 250, 350, 550 \text{ MPa}$. From the observation of these two figures, one readily concludes that:

- (i) Both the slender and stocky columns have stress ratios $\sigma_{CRD} / \sigma_{CRL}$ and σ_{NLD} / σ_U that always exhibit opposite variations with the cross-section dimensions – this suggests that the $\sigma_{CRD} / \sigma_{CRL}$ value may be used to improve the σ_{NLD} estimates (and also the σ_{NDL} ones, for that matter).
- (ii) The slender column σ_{NLD} estimates are all safe in the first plots, mostly unsafe in the second and almost all unsafe in the third. This “safety drop” is most likely related to the b_f / b_w ratio increase: 0.45-0.55 (first plot), 0.61-0.73 (second) and 0.84-0.84 (third).

Table 3(a). Comparison Between the “Exact” and DSM (σ_{NLD}) Ultimate Strength Estimates (STC1)

	b_f	L	Imp.	f_y	FEA			DSM				σ_{NLD}/σ_U
					σ_{CRL}	σ_{CRD}	σ_U	λ_D	σ_{ND}	λ_{LD}	σ_{NLD}	
$b_w=180\text{mm}, b_s=20\text{mm}, t=3.4\text{mm}$	90	650	D	250	361	399	240	0.79	221	0.78	220	0.92
				350			298	0.94	276	0.87	256	0.86
				550			361	1.17	360	1.00	306	0.85
	95	650	D	250	358	377	231	0.81	218	0.78	217	0.94
				350			287	0.96	270	0.87	252	0.88
				550			341	1.21	351	0.99	300	0.88
	100	650	D	250	355	355	222	0.84	213	0.78	213	0.96
				350			276	0.99	264	0.86	247	0.90
				550			323	1.24	342	0.98	294	0.91
	105	650	D	250	353	338	217	0.86	210	0.77	210	0.97
				350			267	1.02	259	0.86	243	0.91
				550			307	1.28	334	0.97	289	0.94
	110	650	D	250	350	320	211	0.88	206	0.77	206	0.98
				350			256	1.04	253	0.85	239	0.93
				550			292	1.31	326	0.96	284	0.97

Table 3(b). Comparison Between the “Exact” and DSM (σ_{NLD}) Ultimate Strength Estimates (STC2)

	b_f	L	Imp.	f_y	FEA			DSM				σ_{NLD}/σ_U
					σ_{CRL}	σ_{CRD}	σ_U	λ_D	σ_{ND}	λ_{LD}	σ_{NLD}	
$b_t=78\text{mm}, b_s=30\text{mm}, t=2.8\text{mm}$	100	800	D	250	736	656	249	0.62	247	0.58	247	0.99
				350			345	0.73	324	0.66	324	0.94
				550			508	0.92	442	0.77	442	0.87
	105	800	D	250	680	641	249	0.62	246	0.60	246	0.99
				350			344	0.74	322	0.69	322	0.94
				550			503	0.93	438	0.80	429	0.85
	110	800	D	250	630	625	248	0.63	246	0.62	246	0.99
				350			344	0.75	320	0.71	320	0.93
				550			499	0.94	434	0.83	416	0.83
	115	800	D	250	581	611	248	0.64	245	0.65	245	0.99
				350			342	0.76	318	0.74	318	0.93
				550			493	0.95	430	0.86	403	0.82
	120	800	D	250	538	596	248	0.65	244	0.67	244	0.98
				350			341	0.77	316	0.77	316	0.93
				550			489	0.96	426	0.89	391	0.80

Table 3(c). Comparison Between the “Exact” and DSM (σ_{NLD}) Ultimate Strength Estimates (STC3)

	b_f	L	Imp.	f_y	FEA			DSM				σ_{NLD}/σ_U
					σ_{CRL}	σ_{CRD}	σ_U	λ_D	σ_{ND}	λ_{LD}	σ_{NLD}	
$b_w=100\text{mm}, b_t=100\text{mm}, t=2\text{mm}$	22	950	D	250	317	285	226	0.94	197	0.79	195	0.86
				350			262	1.11	241	0.87	224	0.85
				550			276	1.39	308	0.99	265	0.96
	24	950	D	250	317	299	227	0.92	201	0.80	198	0.87
				350			270	1.08	246	0.88	227	0.84
				550			287	1.36	315	1.00	268	0.94
	26	950	D	250	317	314	230	0.89	205	0.80	200	0.87
				350			279	1.06	251	0.89	230	0.83
				550			300	1.32	323	1.01	273	0.91
	28	950	D	250	316	331	232	0.87	208	0.81	202	0.87
				350			288	1.03	257	0.90	233	0.81
				550			316	1.29	331	1.02	277	0.88
	30	950	D	250	315	350	234	0.85	212	0.82	205	0.88
				350			297	1.00	263	0.91	237	0.80
				550			337	1.25	339	1.04	281	0.83

Table 4(a). Comparison Between the “Exact” and DSM (σ_{NDL}) Ultimate Strength Estimates (STC1)

	b_f	L	Imp.	f_y	FEA			DSM				
					σ_{CRL}	σ_{CRD}	σ_U	λ_L	σ_{NL}	λ_{DL}	σ_{NDL}	σ_{NDL}/σ_U
$b_w=180\text{mm}, b_s=20\text{mm}, t=3.4\text{mm}$	90	650	D	250	361	399	240	0.83	239	0.77	215	0.89
				350			298	0.98	301	0.87	251	0.84
				550			361	1.23	406	1.01	302	0.84
	95	650	D	250	358	377	231	0.84	239	0.80	211	0.91
				350			287	0.99	300	0.89	245	0.85
				550			341	1.24	405	1.04	295	0.86
	100	650	D	250	355	355	222	0.84	238	0.82	206	0.93
				350			276	0.99	299	0.92	240	0.87
				550			323	1.24	404	1.07	287	0.89
	105	650	D	250	353	338	217	0.84	238	0.84	203	0.93
				350			267	1.00	298	0.94	235	0.88
				550			307	1.25	403	1.09	281	0.91
	110	650	D	250	350	320	211	0.84	237	0.86	199	0.94
				350			256	1.00	298	0.96	230	0.90
				550			292	1.25	402	1.12	274	0.94

Table 4(b). Comparison Between the “Exact” and DSM (σ_{NDL}) Ultimate Strength Estimates (STC2)

	b_w	L	Imp.	f_y	FEA			DSM				
					σ_{CRL}	σ_{CRD}	σ_U	λ_L	σ_{NL}	λ_{DL}	σ_{NDL}	σ_{NDL}/σ_U
$b_f=78\text{mm}, b_s=30\text{mm}, t=2.8\text{mm}$	100	800	D	250	736	656	249	0.58	250	0.62	247	0.99
				350			345	0.69	350	0.73	324	0.94
				550			508	0.86	514	0.88	423	0.83
	105	800	D	250	680	641	249	0.61	250	0.62	246	0.99
				350			344	0.72	350	0.74	322	0.94
				550			503	0.90	501	0.88	412	0.82
	110	800	D	250	630	625	248	0.63	250	0.63	246	0.99
				350			344	0.75	350	0.75	320	0.93
				550			499	0.93	489	0.88	402	0.81
	115	800	D	250	581	611	248	0.66	250	0.64	245	0.99
				350			342	0.78	350	0.76	318	0.93
				550			493	0.97	476	0.88	392	0.80
	120	800	D	250	538	596	248	0.68	250	0.65	244	0.98
				350			341	0.81	342	0.76	311	0.91
				550			489	1.01	464	0.88	383	0.78

Table 4(c). Comparison Between the “Exact” and DSM (σ_{NDL}) Ultimate Strength Estimates (STC3)

	b_s	L	Imp.	f_y	FEA			DSM				
					σ_{CRL}	σ_{CRD}	σ_U	λ_L	σ_{NL}	λ_{DL}	σ_{NDL}	σ_{NDL}/σ_U
$b_w=100\text{ mm}, b_f=100\text{ mm}, t=2\text{ mm}$	22	950	D	250	317	285	226	0.89	230	0.90	187	0.83
				350			262	1.05	288	1.00	215	0.82
				550			276	1.32	388	1.17	256	0.93
	24	950	D	250	317	299	227	0.89	230	0.88	190	0.84
				350			270	1.05	288	0.98	219	0.81
				550			287	1.32	388	1.14	261	0.91
	26	950	D	250	317	314	230	0.89	229	0.85	193	0.84
				350			279	1.05	288	0.96	223	0.80
				550			300	1.32	388	1.11	267	0.89
	28	950	D	250	316	331	232	0.89	229	0.83	197	0.85
				350			288	1.05	288	0.93	228	0.79
				550			316	1.32	388	1.08	273	0.86
	30	950	D	250	315	350	234	0.89	229	0.81	200	0.86
				350			297	1.05	287	0.91	232	0.78
				550			337	1.32	387	1.05	279	0.83

- (iii) While in the slender columns with narrow flanges (first plot) a f_y increase leads to a σ_{NLD} / σ_U drop (i.e., to safer but less accurate estimates), precisely the opposite behaviour is exhibited by the slender columns with moderate-to-wide flanges (second and third plots): a higher f_y leads to a σ_{NLD} / σ_U increase (i.e., to more unsafe results). Thus, the DSM estimate accuracy always drops as f_y increases.
- (iii) While in the slender columns with narrow flanges (first plot) a f_y increase leads to a σ_{NLD} / σ_U drop (i.e., to safer but less accurate estimates), precisely the opposite behaviour is exhibited by the slender columns with moderate-to-wide flanges (second and third plots): a higher f_y leads to a σ_{NLD} / σ_U increase (i.e., to more unsafe results). Therefore, it seems logical to assume that the DSM estimate accuracy always drops as the yield stress f_y increases.
- (iv) The stocky column σ_{NLD} estimates are always safe in all the three plots – unlike in the slender columns, an increase of the b_f/b_w ratio does not lead to unsafe estimates.
- (v) In stocky columns with moderate flanges (second plot), a f_y increase leads to a σ_{NLD} / σ_U drop (i.e., to safer but less accurate results). Conversely, in stocky columns with narrow or wide flanges (first and third plots), there is no visible tendency, as far as the influence of f_y on the safety level of the σ_{NLD} estimates is concerned.

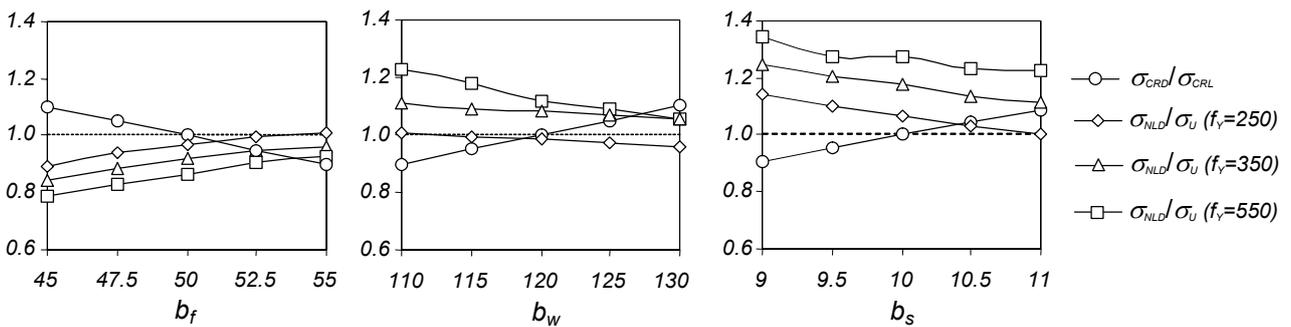


Figure 9. Slender Columns: Variation of $\sigma_{CRD} / \sigma_{CRL}$ and σ_{NLD} / σ_U Ratios with the Cross-Section Dimensions

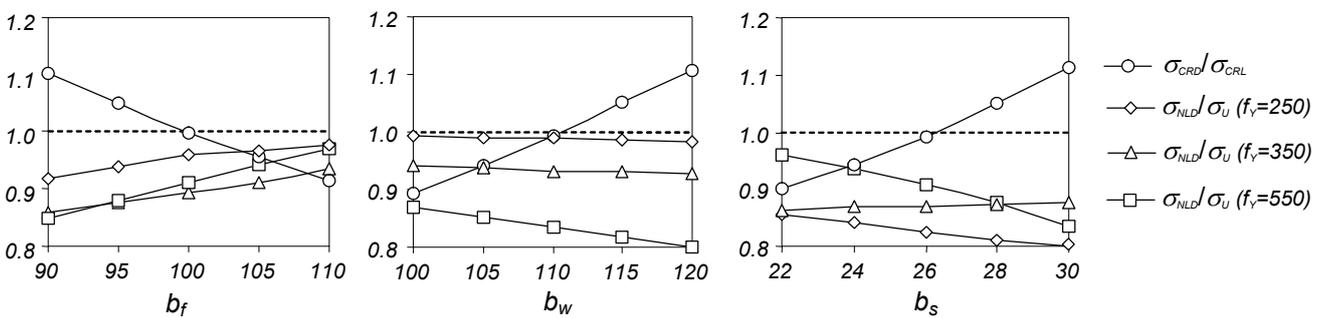


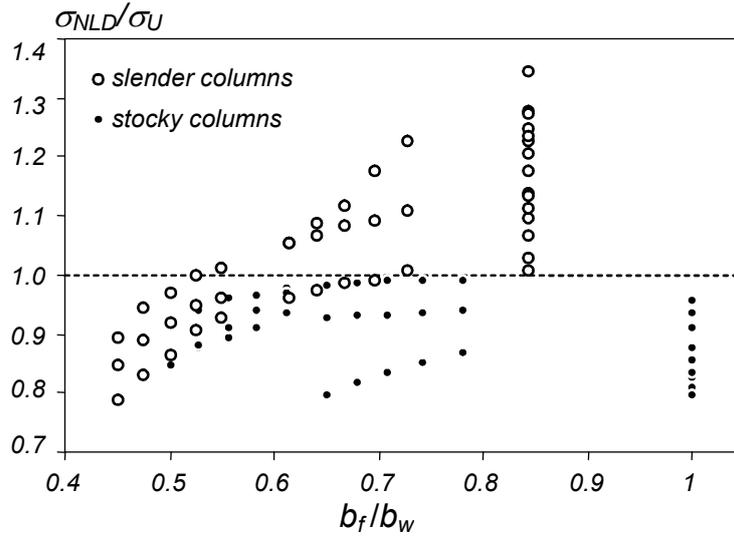
Figure 10. Stocky Columns: Variation of $\sigma_{CRD} / \sigma_{CRL}$ and σ_{NLD} / σ_U Ratios with the Cross-Section Dimensions

Figures. 11(a)-(b) show the variation of σ_{NLD} / σ_U with (i) the flange-to-web width ratio b_f/b_w ratio and (ii) the yield stress f_y ($f_y=250, 350, 550 \text{ MPa}$). These plots provide the following information:

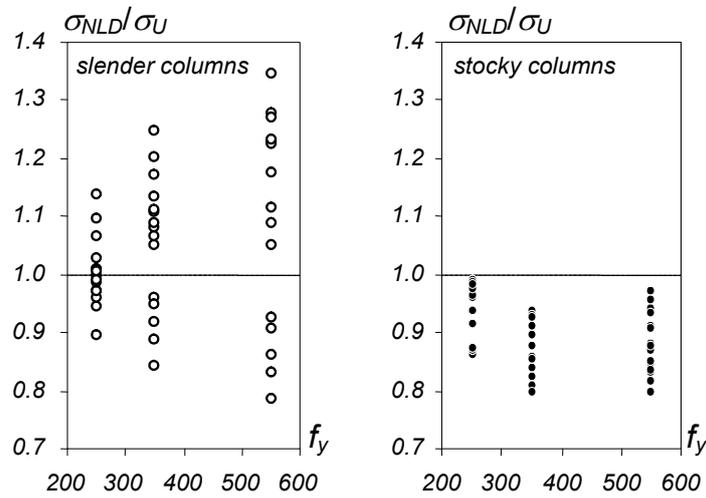
- (i) For the slender columns, it is clear that an increase in b_f/b_w always leads to an increase in the value of the stress ratio σ_{NLD}/σ_U . Moreover, the σ_{NLD} estimates are (i₁) mostly safe for columns with narrow flanges ($b_f/b_w < 0.6$) and (i₂) mostly unsafe for columns with moderate-to-wide flanges ($b_f/b_w > 0.6$).
- (ii) For the stocky columns, a b_f/b_w increase does not cause a similar increase in the stress ratio σ_{NLD}/σ_U .
- (iii) In slender columns, a f_y increase leads to a spreading of the σ_{NLD}/σ_U values around (above and below) 1.0 – see Figure. 11(b). In stocky columns, on the other hand, a higher yield stress does not alter significantly the scatter of the σ_{NLD}/σ_U values.
- (iv) In columns with moderate-to-wide flanges, the LPM is triggered by the flange (not the web) instability. Since the DM is always caused by the flange-stiffener/lip instability, the flange local-plate deformations will certainly have a deteriorating effect on the column distortional post-buckling behaviour.
- (v) Whenever the yield stress f_y is much higher than the σ_{CRD} and σ_{CRL} values (this is the case in the slender columns with moderate-to-wide flanges and $f_y = 550 \text{ MPa}$), there is lots of “room” for the local-plate/distortional mode interaction to develop before the applied stresses reach their ultimate value σ_U . This feature will certainly have a weakening impact on the ultimate strength of the columns affected by LP/D mode interaction.

Finally, Figure. 12 shows the variation of the σ_U/f_y (white dots) and σ_{NLD}/f_y (black dots) stress ratios with the columns *distortional slenderness* $\lambda_D = (f_y/\sigma_{CRD})^{0.5}$, for the three yield stress values. Also included are the two DSM “Winter-type” curves providing the *local-plate* and *distortional* column ultimate strengths, which were defined in Eqs. 3 and 4. The joint observation of all these results prompts the following comments:

- (i) The proposed DSM predictions (σ_{NLD}) concerning the *slender* columns ($\lambda_D > 1.4$) always lie well below both the local-plate and distortional DSM curves. On the other hand, these same predictions for the *stocky* columns ($\lambda_D < 1.4$) are always located near the distortional DSM curve. This means that, at least for the critical stress ratio range considered ($0.90 \leq \sigma_{CRD}/\sigma_{CRL} \leq 1.10$), the local-plate/distortional interaction always causes a substantial strength erosion in the *slender* columns (with respect to the values associated with the individual local-plate and distortional collapses).
- (ii) Unlike the σ_{NLD} estimates, which always remain quite “aligned” (*i.e.*, define a single curve), the “exact” ultimate strengths exhibit a “vertical dispersion” that grows with f_y . This trait indicates that, as f_y increases, the column σ_U values become “more dependent” on the value of the slenderness λ_D .



(a)



(b)

Figure 11. Variation of the Stress Ratio σ_{NLD} / σ_U with (a) b_f/b_w and (b) f_y

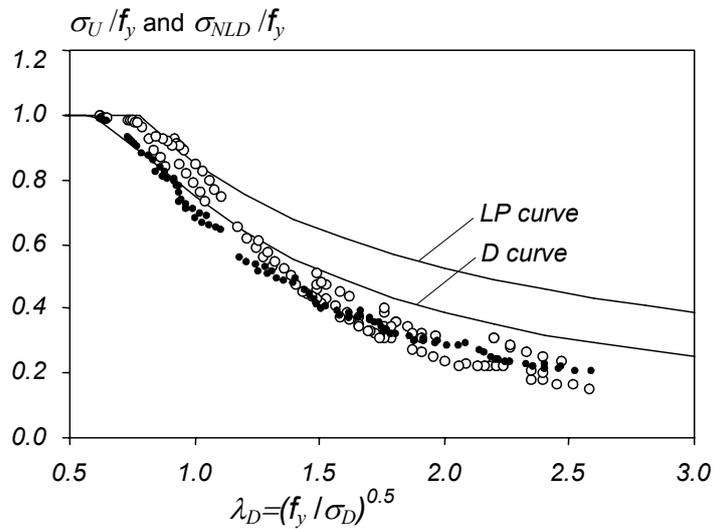


Figure 12. Variation of σ_{NLD} / f_y and σ_U / f_y with the Distortional Slenderness λ_D , Plus the LP and D DSM Curves

5. CONCLUSION

The results of an ongoing investigation concerning the use of a DSM (Direct Strength Method) approach to estimate the ultimate strength of lipped channel columns affected by local-plate/distortional mode interaction were reported. The columns analysed (a total of 111) were all simply supported and had either low-to-moderate ($0.6 \leq \lambda_D \leq 1.4$ – *stocky* columns) or moderate-to-high ($1.4 \leq \lambda_D \leq 2.6$ – *slender* columns) distortional slenderness – therefore, they covered the slenderness range inside which the individual local-plate and distortional DSM curves were experimentally and/or numerically calibrated and validated. On the basis of the ultimate load values obtained by means of a FEM-based parametric study, it was possible (i) to assess the performance of two basic DSM approaches that are based on the already well established expressions derived to estimate the ultimate strengths of columns failing in individual (“pure”) local-plate and distortional modes, and also (ii) to identify a number of features that must be included in a more elaborate DSM approach, specifically developed to take into account the local-plate/distortional buckling mode interaction phenomenon.

Concerning the use of the two basic DSM approaches (*NDL* and *NLD*) to predict the ultimate strength (σ_U) of columns experiencing strong local-plate/distortional mode interaction (the local-plate and distortional critical stresses, σ_{CRL} and σ_{CRD} , are never more than 10% apart), it was possible to conclude that:

- (i) In columns with low-to-moderate distortional slenderness ($\lambda_D \leq 1.4$), the σ_{ND} values provide accurate estimates of their load-carrying capacities – *i.e.*, the use of the existing DSM provisions for distortional buckling yields satisfactory results.
- (ii) For moderate-to-high column distortional slenderness ($1.4 \leq \lambda_D \leq 2.6$), the σ_{NLD} values provide reasonably accurate ultimate strength estimates – in particular, the “quality” of these estimates is slightly superior to that of their σ_{NDL} counterparts. However, it was noticed that the σ_{NLD} values consistently overestimated the σ_U ones in the columns exhibiting wide flanges and high yield stresses.
- (iii) Nevertheless, it is fair to say that, at least for the simply supported columns dealt with in this study, the σ_{NLD} values provide much better column ultimate strength predictions than the existing DSM provisions (for pure local-plate and distortional failures). Indeed, these values are mostly safe and fairly accurate, regardless of the distortional slenderness value, which confirms the assessment made by Yang and Hancock [10], on the basis of an experimental investigation involving lipped channel columns with “v-shape” web and flange intermediate stiffeners.

In spite of the fairly good performance of the *NLD* DSM approach to account for the local-plate/distortional mode interaction effects, the authors of this paper (i) are aware that further investigations are required (*e.g.*, one must analyse fixed columns and additional experimental evidence is necessary) and also (ii) feel that there is still room for improvement, particularly for the slender columns that exhibit wide flanges and high yield stresses, *i.e.*, the ones for which the σ_{NLD} predictions excessively overestimate the “exact” σ_U values. Therefore, the ultimate goal of the research effort currently under way is to develop, validate and calibrate (through the comparison with additional numerical simulations and experimental results) a direct strength approach that is able to cover adequately all columns affected by local-plate/distortional interactive buckling. Hopefully, this goal will be achieved for lipped channel columns in a not too distant future – some fixed column numerical results were very recently reported [24, 25] and an experimental test program is currently being carefully programmed.

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